

EDA122/DIT061 Fault-Tolerant Computer Systems

Welcome to Lecture 7

Generalized Stochastic Petri-Nets (GSPNs)

Design Diversity in the
Airbus A330/A340 Fly-by-wire system

Outline

- Generalized Stochastic Petri Nets (GSPNs)
 - Availability GSPN model of hot standby systems
 - Reachability graph
 - Elements of GSPN:s
 - Examples: construction of GSPN models for various systems
- Design diversity in Airbus A330/A340 fly-by-wire system

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2

Generalized Stochastic Petri Nets (GSPN)

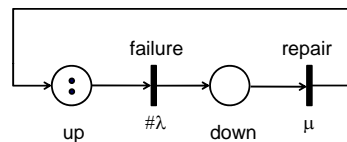
- A GSPN provides a graphical syntax for specifying state space models (Markov models)
- It provides a more compact way of describing a state space model than a state diagram
- A Petri net consists of
 - Places (circles)
 - Transitions (vertical bars)
 - Arcs (arrows)
 - Tokens (dots)

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3

GSPN modell of repairable hot standby system with one spare units



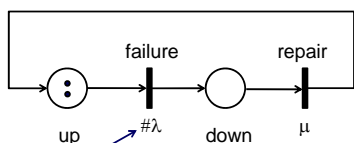
Marking shows the case when both modules are working: there are two tokens in the place "up"

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4

GSPN modell of repairable hot standby system with one spare unit



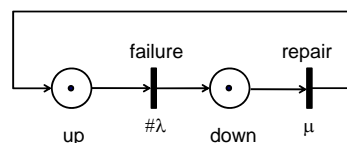
The # character indicates that the firing rate depends on the number of tokens in the place "up"

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5

GSPN modell of repairable hot standby system with one spare unit



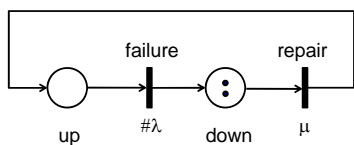
Marking when one module is up and one is down

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6

GSPN modell of repairable hot standby system with one spare unit



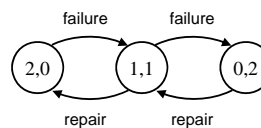
Marking when both modules are down

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7

Reachability Graph for the GSPN model



State labelling:

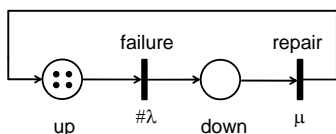
(X, Y) X = Number of tokens in "up"
Y = Number of tokens in "down"

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8

GSPN modell of repairable hot standby system with 3 spare units



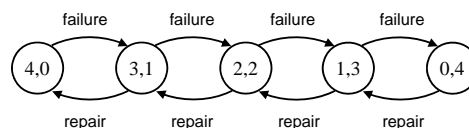
Marking when all modules are working (= four tokens in place "up")

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9

Reachability Graph for hot standby system with 3 spares



State labelling:

(X, Y) X = Number of tokens in "up"
Y = Number of tokens in "down"

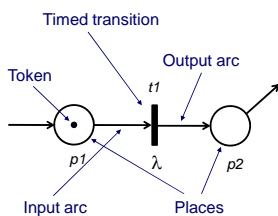
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10

Elements of GSPNs

- Places – holds tokens
- Transitions – moves tokens from one place to another
- Arcs – connects transitions with places
- Tokens – moves between places via transitions
- Marking – a certain placement of tokens in the Petri net.



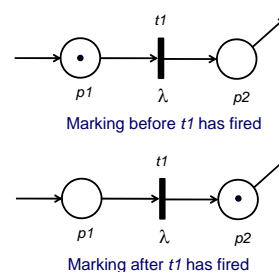
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11

Timed Transitions in GSPNs

- Timed transitions are drawn as a *thick* vertical line
- The timed transition $t1$ fires at a random point in time after a token has arrived in $p1$
- The firing time is exponentially distributed with the rate λ
- When $t1$ fires, one token moves from $p1$ to $p2$
- In this example, the firing rate is constant, i.e., independent of the number of tokens in $p1$.



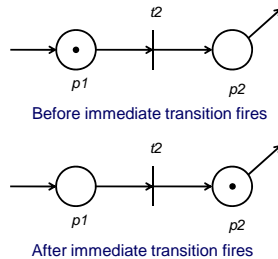
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12

Immediate Transition in GSPNs

- An immediate transition is drawn as a *thin* vertical line.
- t_2 fires immediately when one token has arrived in p_1



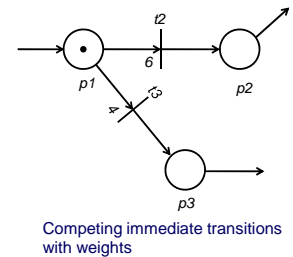
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13

Weights for Immediate Transitions

- Immediate transitions leaving the same states can be assigned weights
- In this example, t_2 has a weight of 6 and t_3 has a weight of 4, which means that t_2 fires with 60% probability and t_3 fires with 40% probability when a token enters p_1
- Note: Weights are normalized to sum up to one when the GSPN is analysed



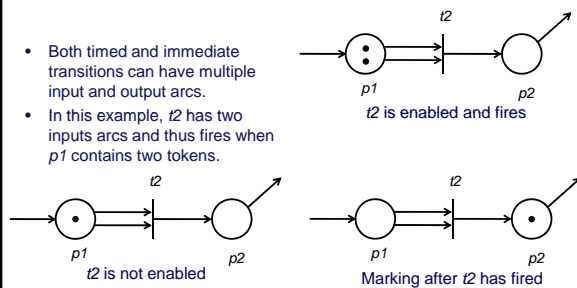
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14

Arc Multiplicity

- Both timed and immediate transitions can have multiple input and output arcs.
- In this example, t_2 has two inputs arcs and thus fires when p_1 contains two tokens.



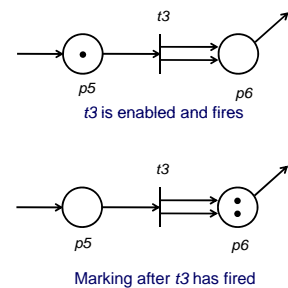
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Multiple Output Arcs

- In this example, t_3 has two outputs arcs and thus produces two tokens when it fires.



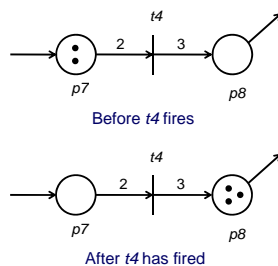
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16

Simplified notation for multiple arcs

- The number of input and output arcs can be given by the number placed just above an arc.
- In this example, t_4 has 2 input arcs and 3 output arcs.



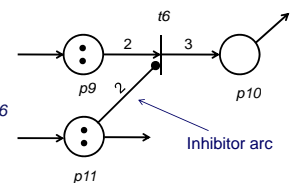
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17

Inhibitor arcs

- An inhibitor arc blocks the firing of a transition based on the marking of a place
- If p_{11} has 2 or more tokens the inhibitor arc blocks the firing of t_6



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18

Problems

1. Construct a GSPN model for calculating the reliability of a system consisting of two modules operating in active redundancy.
2. Construct a GSPN model for calculating the reliability of a TMR system
3. Construct a GSPN model for calculating the reliability of a k -of- n system

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19

GSPN reliability model for system with two modules operating in active redundancy

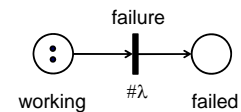


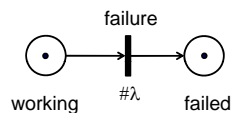
Figure shows GSPN model with initial marking, i.e., with two working modules

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20

GSPN reliability model for system with two modules operating in active redundancy



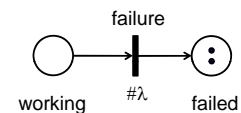
Marking corresponding to one working and one failed module

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21

GSPN reliability model for system with two modules operating in active redundancy



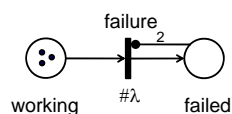
Marking corresponding to system failure

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22

GSPN reliability model for TMR system



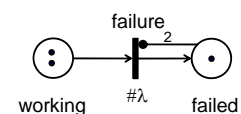
Marking corresponding to three modules working

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23

GSPN reliability model for TMR system

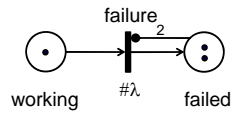


Marking corresponding to two modules working, one module failed

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24

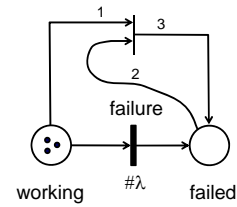
GSPN reliability model for TMR system

Marking corresponding to one module working, two modules failed
Timed transition is disabled by inhibitor arc

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25

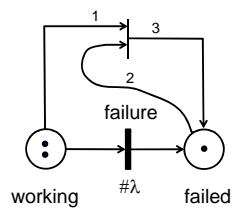
GSPN reliability model for TMR system

Three modules working

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26

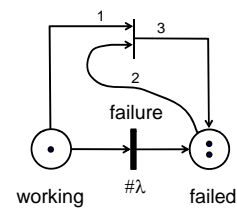
GSPN reliability model for TMR system

Two modules working, one module failed

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27

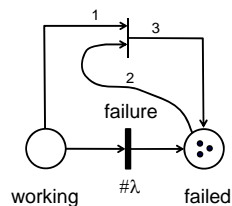
GSPN reliability model for TMR system

One module working, two modules failed => immediate transition enabled

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28

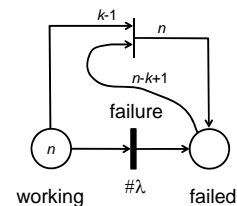
GSPN reliability model for TMR system

Marking after immediate transition has fired – corresponds to system failure

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29

GSPN reliability model for k-of-n system

Marking corresponding to n modules working

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30

Fault tolerance in the Airbus A330/A340 fly-by-wire system

- Motivation
- System overview
- Design diversity

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31

Motivation for fly-by-wire system

- Improving safety through automated control
 - Reducing the pilot's workload
 - 60% of air traffic accidents are due human errors of some kind (not only pilots errors).
 - Reduced workload for the pilot increases safety
 - Prevent the pilot from inadvertently exceeding the aircraft's controllability

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32

Flight control surfaces of A340

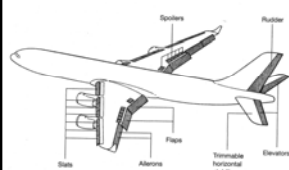


Figure 15.9 The right control surfaces of an A340.

- Primary flight control surfaces
 - Ailerons - controls the roll axis
 - Elevators - controls the pitch axis
 - Ruder - controls the yaw axis
- Secondary flight control surfaces
 - Flaps - lowering the flaps increases drag (air resistance) and lift
 - Spoilers increases drag and reduces lift
 - Slats - prevents stalls

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33

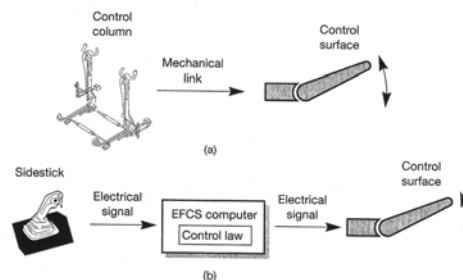


Figure 15.12 A comparison of (a) mechanical and (b) electrical control.

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34

Design Diversity in Airbus A330/A340

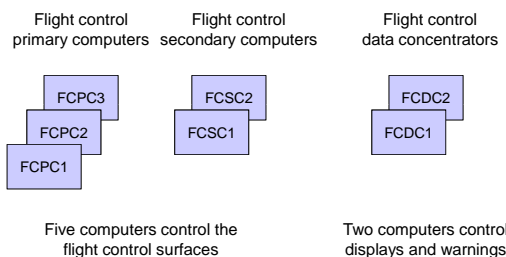
- Two types of computers
 - 3 primary computers
 - 2 secondary computers
- Each computer are internally duplicated and consists of two channels
 - Command channel
 - Monitor channel

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35

Architecture for A330/A340



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36

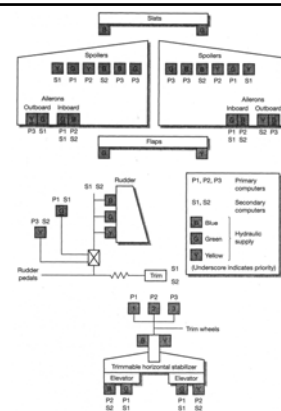
Design Diversity in Airbus A330/A340

- Implementation of primary computers
 - Supplier: Aérospatiale (HW&SW)
 - Hardware: Two Intel 80386 (one for each channel)
 - Software: assembler for command channel, PL/M for monitor channel.
- Implementation of secondary computers
 - Supplier: Sextant Avionique (HW), Aérospatiale(SW)
 - Hardware: Two Intel 80186 (one for each channel)
 - Software: assembler for command channel, Pascal for monitor channel.

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37



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Figure 15.14 The computer and actuator arrangement of the A330/A340.

38

Principle of Graceful Degradation

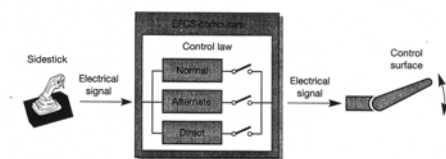


Figure 15.15 The flight control laws.

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39

Features of control laws

- Normal flight control law
 - Stabilization against gusts of wind
 - Pitch trimming
 - Flight envelope protection
 - Prevents stall and overspeed
- Alternate flight control law
 - Pitch trimming and stabilization
 - No flight envelope protection
 - Warns pilot about stall and overspeed conditions
- Direct flight control law
 - The position of the side stick determines the position of the control surfaces.
 - Open-loop control

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40

Normal vs. direct control

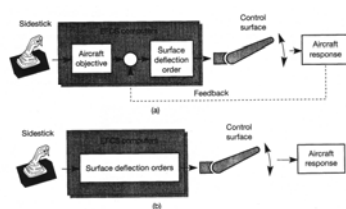


Figure 15.16 A comparison of the normal and direct control laws: (a) normal control law with aircraft feedback; (b) direct control law.

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41

Summary of fault tolerance features in A330/A340

- Mechanical back-up: Mechanical linkages to the rudder and trimmable horizontal stabilizer give control in the event of total electronic system failure
- Computers: Five computers of two types with diverse hardware and software
- Sensors: Dual or triple redundant sensors
- Actuators: Single, double or triple actuators
- Hydraulic supplies: Three independent circuits and five pumps; hydraulic power can be produced by engines and ram air turbine
- Electrical supplies: The A340 uses six generators and two batteries; four generators are driven by the engines, one by a auxiliary power unit (APU) and one by the hydraulic system.

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42

Ram air turbines



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43

Overview of Lecture 8

- Management
- Life-cycles models
- Standards
- Safety case
- Verification and Validation
- Fault-tree analysis
- Failure mode effects analysis

Preparations:

- Lecture notes
- Chapter 3 - 5 in the course book, see reading instructions on home page.

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44

Overview of Lecture 9

- Fault tolerance in space computers
Guest lecture by Torbjörn Hult, RUAG Space Sweden (formerly Saab Space)

Preparations:

- Ariane 501 failure report
- The US space shuttle's computer system, page 152 -154 in the course book
- Lecture slides

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45